

FINAL EXAMINATION

(CE130-1 Mechanics of Materials)

Problem 1: (10 points)

A pin-jointed 3-column structure is shown in the Figure 1. There is an external force, P , acting on the point C . When external force P is below a critical load, P_{cr} , the position shown in the figure is the initial equilibrium position. (1) What are the internal transverse forces for column AC , BC , and CD when $P < P_{cr}$, why? (2) What are the internal moment for column AC , BC , and CD when $P < P_{cr}$, why? (3) Find internal axial forces for column AC , BC , and CD ; (4) Find the critical load P_{cr} ?

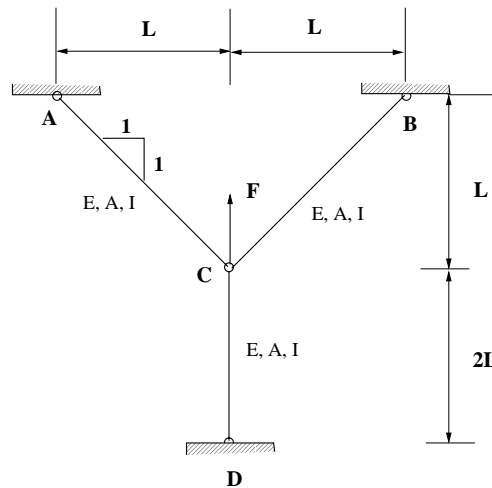


Figure 1: Schematic illustration of problem 1

(Hint: (1) use Castigliano's second theorem, the energy for axially deformed column is, $U = \frac{P^2 L}{2EA}$, where P is the axial force, L is the length of the column, E is the Young's modulus, and A is the cross section of the column; (2) Euler's formula $P_{cr} = \pi^2 EI / L^2$.)

Problem 2 (10 points)

A simply supported beam subjected two concentrated forces that have the same magnitude, P , shown in Figure 2. Draw shear and moment diagrams. (10 points)

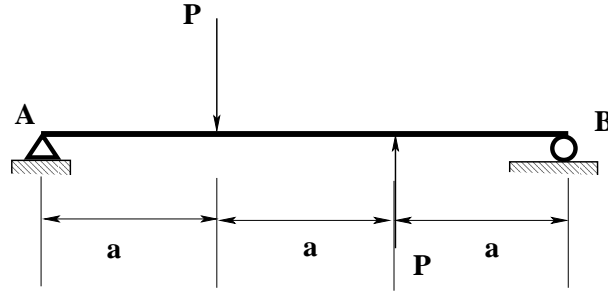


Figure 2: Simply supported beam with concentrated forces.

Problem 3: (20 points)

Consider a plane stress state as follows

$$\boldsymbol{\sigma} = \begin{pmatrix} 5 & 3 \\ 3 & -3 \end{pmatrix} \quad (\text{MPa})$$

- A. draw Mohr's circle of the stress state at that point;
- B. find principal stresses σ_1 , σ_2 , and show the results on properly oriented element in physical space;
- C. find the maximum shear stress, and show the results on properly oriented element in physical space;
- D. find at least one angle between the right face of the initial infinitesimal element and the planes on which the normal stress is zero.

(Hint:

$$\begin{aligned} \sigma_1 &= \frac{\sigma_x + \sigma_y}{2} + \sqrt{\left(\frac{\sigma_x - \sigma_y}{2}\right)^2 + \tau_{xy}^2} \\ \sigma_2 &= \frac{\sigma_x + \sigma_y}{2} - \sqrt{\left(\frac{\sigma_x - \sigma_y}{2}\right)^2 + \tau_{xy}^2} \\ \tan 2\theta_p &= \frac{\tau_{xy}}{(\sigma_x - \sigma_y)/2} \\ \tan 2\theta_s &= -\frac{(\sigma_x - \sigma_y)/2}{\tau_{xy}} \end{aligned}$$

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Problem 4 (10 points)

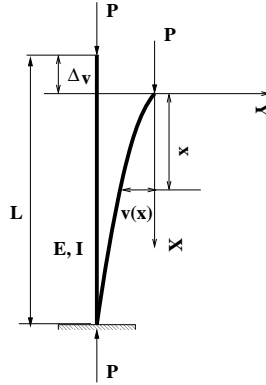


Figure 3: Problem 4

Determine the critical buckling load P_{cr} for a cantilever elastic column with span L and constant stiffness (rigidity) EI .

Use any methods that you feel comfortable with.

(Hint:

(1) Make a cut at the cross section X; Draw free-body diagram for the isolated part, and derive the second order differential equation that governs the stability of a cantilever beam, and find the critical load;

or

(2) Use the fourth order differential equation

$$\frac{d^4 v}{dx^4} + \lambda^2 \frac{d^2 v}{dx^2} = 0 \quad (1)$$

where $\lambda^2 = \frac{P}{EI}$. Write down the boundary conditions, solve the differential equation, and find the critical load. The general form of homogeneous solution of Eq. (1) is $v(x) = C_1 \cos \lambda x + C_2 \sin \lambda x + C_3 x + C_4$.

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Problem 5 (10 points)

Consider an infinitesimal element shown in Figure 4. The normal stresses and shear stresses on two oblique planes are given. Find σ_x and τ_{xy} .

Problem 6 (20 points)

A box beam is made by nailing together four boards in the configurations shown in Figure 5 and labeled as *Config.1* and *Config.2*. The beam supports a concentrated load of 1000 N at its midspan, and it rests on simple supports as shown in Figure 5. Assume each nail can withstand an allowable shear force of 200 N.

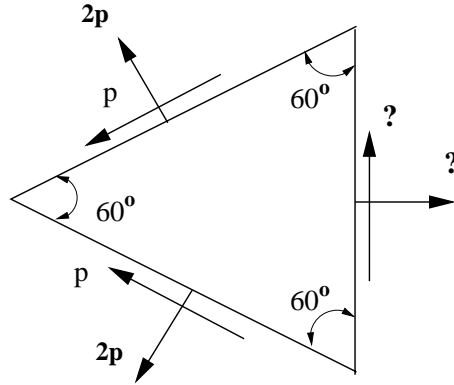


Figure 4: Problem 5

- a . draw shear diagram;
- b . what is the maximum spacing (Δ_s) for configuration 1;
- c . what is the maximum spacing (Δ_s) for configuration 2;

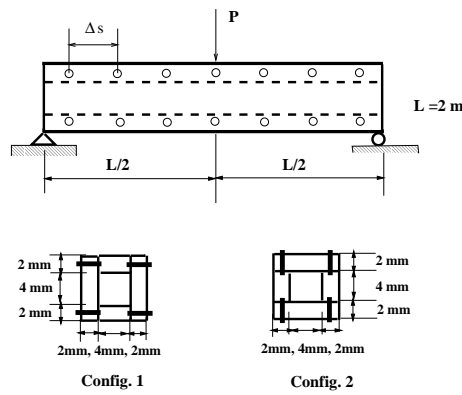


Figure 5: Problem 6

(Hint:

$$q = \frac{VQ}{I_z};$$

$$Q = \int_A y dA = \bar{y}A$$

$$q = \frac{N_{allowable}}{\Delta_s}$$

where $N_{allowable}$ stands for allowable shear force by the nails.)

Problem 7 (10 points)

A planar frame ABCD is subjected a concentrated moment, M , at the end point D as shown in Figure 6.

- (a) draw moment diagram along the frame;
- (b) find the cross section rotation at point D, i.e. θ_D ?

(Hint:

- (1) use virtual force method,

$$\bar{1} \times \theta_D = \int \bar{m}_c(s) \frac{M(s)}{EI} ds$$

where s is a local coordinate;

- or (2) use Castigliano's second theorem,

$$\theta_D = \frac{\partial U^*}{\partial M} = \frac{\partial U}{\partial M}$$

$$U = \frac{1}{2EI} \int_0^L M(s)^2 ds$$

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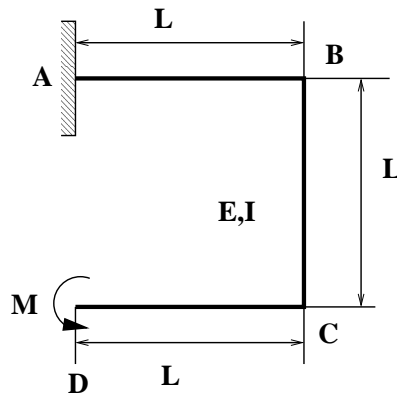


Figure 6: Problem 7

Problem 8 (10 points)

A L-shaped beam is made of a rectangular section and a solid cylinder section with radius $R = 0.1m$. The span of the both section is $L = 2.0 m$. There is a concentrated load, $P = 300N$, acting on the free-end of the rectangular (as shown in Figure 7.). (1) Draw the moment diagram, shear diagram, and internal torque diagram; (2) Find the normal stress σ_x , shear stresses τ_{xy} and

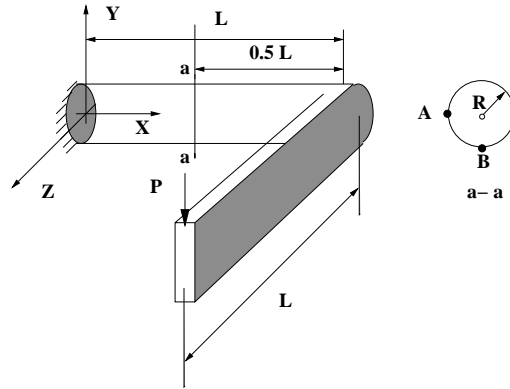


Figure 7: Problem 8

τ_{xz} at point A; (3) Find the normal stress σ_x and shear stress τ_{xy} and τ_{xz} at point B;

Hints:

$$\sigma_x = -\frac{M_z y}{I_z}$$

$$\tau = \frac{VQ(y)}{I_z t}$$

$$\tau = \frac{T\rho}{I_p}$$

$$I_z = \frac{1}{2}I_p = \frac{R^4\pi}{4}$$

For semi-circle,

$$Q(y) \Big|_{y=0} = \frac{2}{3}R^3$$